

High-temperature geopolymer & C_f reinforced composites and their derived CMC

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CONTENTS

1. Introduction

- 2. Thermal evolution and crystallization of geopolymer
- 3. Effects of high-temperature heat treatment on the mechanical properties of C_f /geopolymer composites
- 4. Effects of Sol-SiO₂ impregnation on the ambient and high temperature mechanical properties of C_f /leucite composites
- **5.** Conclusions



1. Introduction

Geopolymer

Characteristics

• Environmental friendly -No CO₂ emission • Low temperature fabrication —RT~150 ℃ • Ceramic formation ability upon heating —improved properties • Adjustable thermal properties -Melting point、CTE

Applications and potential applications

- Geopolymer cement and concrete
- Refractory items
- Preparation of ceramics, i.e. Leucite
- Fireproof high-tech applications: aircraft interior, thermal insulators for racing car engine, etc. as an alternative to organic polymers



short fiber

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Fig.1 Stress-displacement curves of bidirectional carbon fiber reinforced geopolymer (Si/Al=27) composites. Fiber volume is 50%

---- J. Giancaspro, P. N. Balaguru, R. E. Lyon. Proceedings of the Thirteenth International Offshore and Polar Engineering Conference, Honolulu, Hawaoo, USA. 2003





Fig.2 3-points flexural strength of geopolymer (Si/Al=3) composites reinforced by different kind of fibers (fiber contents is about 40vol%).

---- J. Davidovits, M. Davidovits. 36th Annual SAMPE Symposium, 1991, 1939~1949



2. Thermal evolution and crystallization of geopolymer

2.1 Thermal evolution and crystallization mechanism of KGP2.2 Effect of Cesium substitution on the crystallization of geopolymer2.3 Effects of Cesium substitution on the phase transition of leucite



2.1 Thermal evolution and crystallization of KGP



---- Peigang He, Dechang Jia. Ceramics International. 2011, 37 (1), 59-63.





Fig.2-3 XRD pattern of KGP soaked at different temperature for 2h





Fig.2-4 Microstructure of KGP soaked at different temperature for 2h. (a), (b) 800℃; (c), (d) 900℃; (e), (f) 950℃; (g), (h) 1000℃.





Fig.2-5 Fractographs of KGP soaked at different temperature for 2h (a), (b) RT; (c), (d) 800°C; (e), (f) 1000°C

---- Peigang He, Dechang Jia. Ceramics International. 2011, 37 (1), 59-63.



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Fig.2-6 Fracture surface morphology of the KGP treated at different temperature and etched in 3wt.% HF at room temperature for 30s: (a) 800℃, (b) 900℃, (c) 950℃, (d) 1000℃





Fig.2-7 TEM images of: metakaolin (a) and the resulted geopolymer (b)





Fig. 2-8 TEM observation of geopolymer treated at 1000° C for 2h: (a) low magnification, (b) and (c) high magnification, (d) SAD pattern of area D





Fig. 2-9 Crystal structures of leucite: (a) cubic phase, (b) tetrahedral phase

Black tetrahedron : The fourfold rings





Fig. 2-10 DTA of KGP using different heating rates Fig. 2-11 Kissinger plot of $\ln(\beta/T_p^2)$ vs. T_p^{-1}



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Fig. 2-12 SEM micrographs of leucite grains (a) and size distribution (b)



Young's Fracture Flexural Vickers hardness strength modulus toughness Specimen (MPa \cdot m^{1/2}) (MPa) (GPa) (GPa) KGP 12.3 ± 1.2 10.3 ± 1.2 0.2 ± 0.04 0.68 ± 0.04 Leucite ceramic 70.0 ± 6.8 65.0 ± 6.3 1.3 ± 0.16 7.39 ± 0.24 2.0**High CTE** 1.8 1.6 % 380°C−500°C Dilatometry (dL/L⁰), 1.4 \leftrightarrow Great residual stress 1.2 HOW? 1.0 from thermal mismatch 0.8 0.6 Cesium substitution 0.4 $\alpha_{25-1200} = 15.4 \times 10^{-6} ^{\circ} \text{C}$ 0.2 0.0 1000 200 400 600 800 1200 0 Temperature (°C)

Table 2-1 Mechanical properties of the geopolymer and the resulted leucite ceramic

Fig.2-13 Thermal expansion of the ceramics derived from KGP

2.2 Effect of Cesium substitution on the crystallization of geopolymer







Fig.2-16 Thermal shrinkage of $Cs_x K_{(1-x)}GP$





Fig. 2-17 Onset and end temperature of stage IV and crystallization temperature of $Cs_xK_{(1-x)}GP$





Fig. 2-18 XRD patterns of $Cs_x K_{(1-x)}$ GP ceramics heat treated at 1200°C for 2h



2.3 Effects of Cesium substitution on the phase transition of leucite



Fig. 2-19 Slow step-scan XRD patterns in a 2θ range of 24.5~28° (a) and the calculated lattice parameters (b) of leucite





Fig. 2-20 The filling ratio of leucite crystal cell derived from $Cs_x K_{(1-x)}GP$





Fig. 2-21 TEM micrographs of the ceramics derived from $Cs_xK_{(1-x)}GP$: (a) x=0, (b) x=0.1, (c) x=0.2.







2.4 Summary

- Geopolymer technology provides a novel method to fabricate leucite ceramic with greatly improved mechanical properties.
- Thermal shrinkage of geopolymer during heat treatment can be divided into structural resilience, dehydration, dehydroxylation and sintering.
- For the K-based geopolymer, leucite crystallization appeared after the sintering stage and the Avarami parameter indicates the three-dimensional crystal growth mechanism.
- Leucite ceramic posses an adjustable thermal expansion coefficient by doping cesium ions.



3. Effects of high-temperature heat treatment on the mechanical properties of C_f /geopolymer composites

3.1 Effects of heat treatment on the properties of C_f/KGP

3.2 Effects of heat treatment on the mechanical properties of C_f/CsKGP
3.3 Summary





Fig. 3-1 High-temperature heat treatment procedure of the composites



3.1 Effects of heat treatment on the properties of C_f/KGP

- Carbon fiber: Unidirectional
- Matrix: $K_2O \cdot Al_2O_3 \cdot (3 \sim 5)SiO_2 \cdot 11H_2O$
- Composites without and after heat treatment were denoted as: KC-W, KC-1000, KC-1100, KC-1200, KC-1300 and KC-1400, respectively
- ◆ Fiber contents: 20vol% for KC-W and 25vol% for others





Fig. 3-2 XRD patterns of C_f/KGP heat treated at different temperature

---- Peigang He, Dechang Jia. Ceramics International. 2010, 36 (4), 1447-1453.





Fig. 3-3 Weight loss of KGP and C_f/KGP

Fig. 3-4 Total vapor pressure of interfacial reaction of Si–C–O system as a function of temperature ^[32]

$$Si-O + C \longrightarrow SiC+CO(g)$$
 (3-1)

---- Peigang He, Dechang Jia. Ceramics International. 2010, 36 (4), 1447-1453.



Reaction	Standard Gibbs free energy (J·mol ⁻¹)
(1) $C_{(s)} + 0.5O_{2(g)} = CO_{(g)}$	-114400-85.77T
(2) $Si_{(s)} + C_{(s)} = SiC_{(s)}$	-73050+7.66 <i>T</i>
(3) $Si_{(s)} + O_{2(g)} = SiO_{2(s)}$	-907100+175.73 <i>T</i>

Table 3-1 The reactions and standard Gibbs free energy

Gibbs free energy for (3-1): $\Delta G^{\circ} = 605250 - 339.61T$

The theoretical reaction temperature is 1782°C





Fig. 3-5 Thermal shrinkages of the composite heat treated in different directions: (a) perpendicular and (b) parallel to the fiber direction




Fig. 3-6 Morphologies of C_f/KGP composite (a) before and (b) after heat treatment at $1100^{\circ}C$



Fig. 3-7 Microstructure of polished cross section of: (a) KC-1000, (b) KC-1100 (low magnification), (c) KC-1100 (high magnification), (d) KC-1300, (e) KC-1400

---- Peigang He, Dechang Jia. Ceramics International. 2010, 36 (4), 1447-1453.



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Fig. 3-8 Interface morphology of KC-1100: (a) TEM image, (b) and (c) SAD paterns of carbon fiber and matrix in area of B and C



Residual stress:
$$\sigma = \frac{(\alpha_{\rm m} - \alpha_{\rm f}) \times \Delta T E_{\rm f} V_{\rm f}}{V_{\rm f} \times (\frac{E_{\rm f}}{E_{\rm m}} - 1) + 1}$$
 (3-2)

Table 3-1 Thermal residual stress in the composites during cooling process

$\Delta T(^{\circ}\mathbb{C})$	1100	1200	1300	1400
σ_{a} (MPa)	557.5	608.2	658.9	709.6
$\sigma_{\rm r}$ (MPa)	195.5	213.3	231.0	248.8





Fig. 3-9 Interface morphology of KC-1200 (a) and KC-1300°C(b)





Fig. 3-10 Interface morphology of KC-1400

Fig. 3-11 HRTEM micrographs of area B in Fig.3-10: (a) HRTEM image and (b) FFT transformed image









Fig. 3-13 Fractographs of C_f/KGP composites heat treated at different temperature: (a) KC-B, (b) KC-1000, (c) KC-1100, (d) KC-1200, (e) KC-1300, (f) KC-1400

---- Peigang He, Dechang Jia. Ceramics International. 2010, 36 (4), 1447-1453.



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Fig. 3-14 Magnified fractograph of (a) KC-1100, (b) KC-1400





Fig. 3-15 Variations of flexural strength and work of fracture of C_f/KGP composites heat treated at different temperatures



Samples	Flexural strength (MPa)	Work of fracture (J·m ⁻²)	Fracture toughess (MPa·m ^{1/2})	Elastic modulus (GPa)
Geopolymer	12.3±1.2	_	0.2 ± 0.04	10.3±1.2
Leucite ceramic	70.0±6.8	—	1.3±0.16	65.0±6.3
KC-W	132.9±8.2	3874.5±266.8	—	36.5±3.4
KC-1000	95.6±10.5	2354.8±243.6	—	30.4±4.1
KC-1100	268.9 ± 14.6	5003.4±254.4	7.21±0.87	67.4±4.2
KC-1200	181.7±16.3	4233.4±299.5	6.57±0.64	55.9±4.8
KC-1300	160.3±19.2	2444.6±283.2	4.9±0.57	50.6±3.6
KC-1400	54.6±3.7	366.6±37.5	1.05 ± 0.31	41.5±4.2

Table 3-2 Mechanical properties of geopolymer and C_f/KGP composites after heat treatment





Fig. 3-16 Flexural stress-displacement curves of geopolymer and C_f/KGP composites heat treated at different temperatures



3.2 Effects of heat treatment on the properties of C_f/C_sKGP

Carbon fiber: Unidirectional

• Matrix: $0.8K_2O \cdot 0.2Cs_2O \cdot Al_2O_3 \cdot 5SiO_2 \cdot 11H_2O$

Composites without and after heat treatment were denoted as: CsKC-W, CsKC-1000, CsKC-1100, CsKC-1150, CsKC-1200 and CsKC-1300, respectively

◆ Fiber contents: 20vol% for CsKC-W and ~25vol% for others





Fig. 3-17 Weight retained of KGP and C_f/CsKGP composites





Fig. 3-18 XRD patterns of $C_f/CsKGP$ heat treated at different temperatures





Fig. 3-19 Thermal shrinkages of $C_f/CsKGP$ heat treated at different directions





Fig. 3-20 Interface morphology of C_f/CsKGP heat treated at different temperatures : (a) CsKC-1100, (b) CsKC-1200, (c) CsKC-1300





Fig. 3-21 Variations of the mechanical properties of C_f/CsKGP heat treated at different temperatures:
(a) Flexural strength and work of fracture, (b) Young's modulus and fracture toughness



Samples	Flexural strength (MPa)	Work of fracture (J·m ⁻²)	Fracture toughess (MPa m ^{1/2})	Elastic modulus (GPa)
CsKC-W	129.6±7.5	3659.9±314.7		34.7±3.4
CsKC-1000	98.8±13.4	2158.7±431.5	_	27.7±4.1
CsKC-1100	229.2±16.8	4357.9±267.7	6.79±0.57	61.9±8.4
CsKC-1150	268.0±20.7	5538.4±327.5	7.65 ± 0.66	67.8±6.7
CsKC-1200	278.7±18.9	5977.4±297.1	8.37±0.53	71.4±6.3
CsKC-1300	90.9±7.4	1054.8±106.7	1.26±0.24	31.8±4.2

Table 3-3 Mechanical properties of C_f/CsKGP composites heat treated at different temperature





Fig. 3-22 Flexural stress–displacement curves of $C_{\rm f}/C_{\rm s}KGP$ heat treated at different temperatures





Fig. 3-23 Fractographs of C_f/CsKGP composites heat treated at different temperature: (a), (b) 1100°C, (c), (d) 1200°C and (e), (f) 1300°C

3.3 Summary

- After proper heat treatment, the C_f/geopolymer composite can be directly converted into the carbon fiber reinforced leucite ceramic matrix composites, and mechanical properties could be greatly enhanced.
- The property improvement can be attributed to the densified and crystallized matrix, and the enhanced fiber/matrix interface bonding based on the fine-integrity of carbon fibers.
- ◆ For the composite heat treated at higher temperature(≥1300° C), the mechanical properties lowered substantially and it tended to fracture in a very brittle manner owing to the seriously degraded carbon fibers.



4. Sol-SiO₂ impregnation on the ambient and high temperature mechanical properties of C_f /leucite composites

- 4.1 Microstructure of the composites after repeated impregnation
- 4.2 Room temperature mechanical properties
- 4.3 Non-isothermal oxidation of the composites
- 4.4 High-temperature mechanical properties and fracture mechanism
- 4.5 Summary





Fig. 4-1 Relative density vs. number of impregnation/drying cycles with different sol-SiO₂ concentration



4.1 Microstructure of the composites after repeated impregnation



Fig. 4-2 Relative density of the composites before and after sol-SiO₂ impregnation





Fig. 4-3 XRD patterns of the impregnated composite treated at different temperatures



Before sol-SiO₂ impregnation

After sol-SiO₂ impregnation



Fig. 4-4 Microstructures of the polished cross section of HKC composite:(a), (b) in the axial direction of fibers; (c), (d) in the radial direction Fig. 4-5 Microstructure of the polished cross section of ImHKC composite: (a), (b) in the axial direction of fibers; (c),(d) in the radial direction

---- Peigang He, Dechang Jia. Journal of the European Ceramic Society. 2010, 30 (15), 3053–3061.



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Fig. 4-6 Energy dispersive spectrum of point A in the Fig. 4-5



Fig. 4-7 Microstructure at a higher magnification of a sol-SiO₂ sealed crack





4.2 Room temperature mechanical properties

Fig.4-8 Mechanical properties of (Im)HKC and (Im)CsHKC

Specimen	НКС	ImHKC
Density (g/cm ³)	1.81	2.16
Relative Density (%)	79.0	93.4
Flexural strength (MPa)	268.9 ± 14.6	364.7±24.4
Young's modulus (GPa)	67.4±4.2	78.0±6.9
Fracture toughness (MPa·m ^{1/2})	7.21±0.87	8.84±1.04
Work of fracture (J·m ⁻²)	5003.4±254.4	6527.4±342.1

 Table 4-1 Room temperature properties of HKC and ImHKC

Table 4-2 Room temperature properties of HCsKC and ImHCsKC

Specimen	HCsKC	ImHCsKC
Density (g/cm ³)	1.85	2.28
Relative Density (%)	81.4	92.9
Flexural strength (MPa)	278.7±18.9	376.4±21.3
Young's modulus (GPa)	71.4±3.8	79.2±5.7
Fracture toughness (MPa·m ^{1/2})	8.37±0.53	10.14 ± 0.89
Work of fracture (J·m ⁻²)	5977.4±297.4	7068.6±413.8

4.3 Non-isothermal oxidation behavior



Fig. 4-9 Non-isothermal oxidation weight loss of the HKC and ImHKC composites





Fig. 4-10 Elevated temperature flexural strength of HKC and ImHKC





Fig. 4-11Elevated temperature work of fracture of HKC and ImHKC





Fig. 4-12 Elevated temperature Young's modulus of HKC and ImHKC









Fig.4-14 Flexural stress-displacemen curves at different temperature


Fig. 4-15 Typical morphologies of ImHC specimens after bending tests at (a) RT, (b) 700°C, (c) 900°C, (d) 1100°C and (e) 1200°C





Fig. 4-16 Fractographs near the tensile sides of: (a), (b) ImHKC-R; (c), (d) ImHKC-700; (e), (f) ImHKC-900; (g) ImHKC-1100; (h) ImHKC-1200





Fig. 4-17 Fractographs obtained at room temperature of ImHC after flexural strength test at: (a) RT, (b) 700°C, (c) 900°C, (d) 1100°C.
Samples were prepared by broking-off the ImHC specimens in cryogenic liquid nitrogen and similar positions were select for observation in each sample



Fig. 4-18 Fractographs obtained at room temperature of ImHC after flexural strength test at 1100: (a) low magnification fractograph, (b)-(e) corresponding to the area of A-D in (a)

Degree of Oxidation
$$= \left(1 - \frac{R_d}{R_0}\right) \times 100\%$$

 R_d : Average radius of carbon fiber after test,
 R_0 : Initial radius of carbon fiber.



Fig. 4-19 The degree of fiber oxidation vs. distance from the surface of the two composites



4.4 Summary

- Repeated sol-SiO₂ impregnation was an efficient method to seal the cracks and pores in the heat-treated composites and thus greatly enhance their relative densities and ambient and high temperature mechanical properties.
- The impregnated composites possess the density of 2.15~2.2g·cm⁻³, and showed much high specific strength. Composites before & after impregnation fractured in a non-brittle manner at ambient & high temperatures.
- ♦ Over an elevated temperature range from 700°C to 900°C, the strength of the two composites showed anomalous gains and reached their maximum values at 900°C.
- The increased strength was attributed to the cracks blunting and release of residual strength based on the integrity of the carbon fiber.
- Compared with the composites without sol-SiO₂ impregnation, the impregnated one showed superior high temperature properties, which was attributed to the improved fiber integrity due to its much denser microstructure.



5. CONCLUSIONS

- Geopolymer can be converted into ceramic materials heat treated at proper high temperature range. The resulted ceramics possess adjustable coefficient of thermal expansion, which makes them excellent candidate matrix as structural materials.
- 2 With proper heat treatment, carbon fiber reinforced geopolymer composites can be converted into C_f/Leucite composites, which showed excellent high temperature mechanical properties up to 1200°C.
- ③ Geopolymer technology together with cerium incorporation, hightemperature heat treatment and sol-SiO₂ impregnation etc provides a new method to fabricate a new composites of low cost and high specific strength, which is attractive for elevated temperature applications.



